

SECTION 9

Pilot's Operating Handbook Supplement AS-11

Emergency Locator Transmitter (ELT) KANNAD 406 AF-Compact or KANNAD Integra 406 AF



This supplement is applicable and must be inserted into Section 9 of the POH when the 406 MHz-ELT KANNAD 406 AF-Compact or Integra 406 AF is installed in the AQUILA AT01-100/200. The information in this supplement adds to or replaces information in the basic POH.

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0.1 RECORD OF REVISIONS

Issue	Reason for Change	Effected Pages	Date of Issue
A.01	Initial Issue	All	28.05.2013
A.02	Editorial Changes	All	15.10.2013
A.03	also valid for AT01-200, Integra 406 AF	All	02.03.2020

0.2 LIST OF CURRENT PAGES

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1. GENERAL

This supplement provides a general description of the 406 MHz-ELT KANNAD 406 AF-Compact / INTEGRA 406 AF, its basic operation and integration into the AQUILA AT01-100/200.

For a more detailed description and full operating instructions, please refer to the current issue of the KANNAD Installation Manual/Operation Manual/Inspection Log.

The information contained in this supplement is to be used together with the complete POH. Furthermore, the KANNAD Installation Manual/Operation Manual/Inspection Log must always be carried on board the aircraft during flight.

2. OPERATING LIMITATIONS

The KANNAD ELT is installed as optional equipment whose failure is uncritical under all operational conditions. The operating limitations defined in section 2 of the basic POH apply without any changes or restrictions.

The ELT Remote Control Panel must be installed. Installation is <u>not</u> optional! If the KANNAD ELT is installed in the aircraft, the following placards must be installed at the locations stated below:

1) Placard on the outer surface of the fuselage in the vicinity of the ELT:



2) Placard on the instrument panel, beside the ELT Remote Control Panel:

ELT-REMOTE-CONTROL Switch ELT –Transmitter to ARM !

3) Placard in the upper right section of the instrument panel:

ELT and FireExtinguisher behind co-pilot seat

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3. EMERGENCY PROCEDURES

The KANNAD ELT is installed as optional equipment whose failure is uncritical under all operational conditions. The emergency procedures defined in section 3 of the basic POH apply without any changes or restrictions.

4. NORMAL PROCEDURES

No change to the basic POH. A short description of the operation of the KANNAD 406 AF/AF-Compact ELT is contained in section 7 of this supplement.

5. PERFORMANCE

No change to the basic POH.

6. WEIGHT AND BALANCE

The change in empty weight and the corresponding center of gravity after the installation or removal of the KANNAD ELT must be determined and recorded in accordance with section 6 of the basic POH.

7. SYSTEMS DESCRIPTION

7.1 GENERAL

The Emergency Locator Transmitter (ELT) is an emergency transmitter that, if activated, transmits a signal on the international civilian emergency frequency (121.5 MHz). This enables rescue teams to locate a lost aircraft in a shorter period of time. Furthermore, the ELT transmits digital messages on 406 MHz that can be processed by COSPAS-SARSAT system satellites to aid and coordinate search and rescue (SAR) operations all over the world. Besides processing and relaying signals received on 406 MHz, these satellites also relay signals transmitted on the international civilian emergency frequency 121.5 MHz to one of 64 ground stations within the COSPAS-SARSAT system. Here SAR operations are initiated and coordinated. The signal transmitted on 406 MHz carries data which identifies the aircraft in distress and helps facilitate SAR operations. The aircraft is located using the Doppler Effect with a precision of app. 2 NM at any point on the earth.

The INTEGRA 406 AF in addition also has an internal GPS system and an internal transmission antenna. When this ELT is activated, the digital signal message is broadcasted at 406 MHz with a significantly more precise GPS position and in the event of a faulty external antenna connection the internal reserve transmission antenna is used for transmitting.

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INTEGRATION OF THE KANNAD ELT INTO THE AQUILA AT01-100 / 200

The ELT is installed on the right side of the baggage compartment floor behind the copilot's seat. The ELT antenna is mounted outboard on the upper fuselage skin behind the baggage compartment bulkhead of the aircraft. A remote control panel for the ELT is installed in the right section of the instrument panel above the engine instruments. The ELT is connected to the remote control panel with a separate cable harness which is routed along with the fuselage cable harness through the cockpit. On the ELT side, the cable harness is equipped with a DIN-12 connector and with a D-SUB 9-pin receptacle on the remote control panel side.



KANNAD 406 AF-COMPACT / INTEGRA 406 AF FRONT VIEW

CONTROLS & CONNECTORS

The following controls are found on the ELT front panel (refer to picture above):

- 1. 3-position switch ARM/OFF/ON
- 2. Red light (LED) *
- 3. DIN 12 socket for connection to remote control panel, dongle or programming equipment
- 4. BNC connector for the antenna
- * The LED of the KANNAD INTEGRA 406 AF ELT is located on the top of the unit.

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The red light (LED) gives an indication to the current mode of the beacon:

• After the self-test:

a series of short flashes indicate the self-test found a problem, one long flash indicates that the self-test is OK.

 In operating mode: periodic flashes during 121.5 MHz transmissions and a long flash during 406 MHz transmission.

A buzzer gives an aural indication to the current mode of the beacon:

Self-test

Continuous tone 2 beeps per second

Transmitting on 121,5 MHzTransmitting on 406 MHz

ELT-REMOTE CONTROL PANEL

The ELT-Remote Control Panel (RC200) is installed in the right section of the instrument panel above the engine instruments.

No tone



The following controls are to be found on the remote control panel:

- 3-position switch (ON/ARMED/RESET TEST)
- Red light (LED adjacent to the "ON" marking)



The ELT can only be operated by the remote control panel if the ELT switch is in the "ARM"-Position.

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TRANSMITTER

The KANNAD 406 AF-Compact / 406 Integra AF can be activated either automatically by the shock sensor (when a crash occurs) or manually by the switch on the ELT or with the remote control panel. The ELT is designed to transmit on the following frequencies:

• On the international emergency frequency 121.5 MHz as well as 406 MHz for COSPAS/SARSAT services (KANNAD 406 AF-Compact / 406 Integra AF)

Both international emergency frequencies are primarily used for homing during the final stages of the SAR (Search and Rescue) operations. The 406 MHz frequency is used to pinpoint the location and identify the aircraft in distress using the COSPAS-SARSAT system. Once activated, the transmitter operates continuously on 121.5 MHz with an output power of 100 mW. After activating the transmitter, a digital signal message at 406 MHz is sent to the COSPAS-SARSAT satellites with a transmission power of approximately 5 W with the aircraft identification and the known GPS position data (only INTEGRA 406 AF) until the battery capacity is exhausted.

POWER SUPPLY

The ELT is supplied with electrical power independent from the on board electrical system of the aircraft. The energy supply is provided by a battery pack composed of two (KANNAD 406 AF-Compact and Integra 406 AF) LiMnO₂ D-cells. The capacity of the battery pack allows transmission on 406 MHz for more than 24 h and 121.5 MHz for more than 48 h at -20° C.



The battery pack cannot be recharged!

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The maximum operating time of the battery pack is specified with a printed expiry date. The battery pack must be replaced if one of the following applies:

- a) After the transmitter has been used in an emergency situation (including any inadvertent activation of unknown duration).
- After the transmitter has accumulated more than one hour of operation (e.g. time accumulated in several tests and inadvertent activations of known duration).
- c) On or before the battery expiry date (see mark on the battery pack or on the transmitter unit).

REGISTRATION AND PROGRAMMING

CAUTION

The ELT must be registered with the local registration authority prior to installation in the aircraft. Change of ownership must also be reported to the local registration authority.

For the declaration and registration of the 406 MHz ELT the forms available from the local registration authority must be used. A data sheet to program the ELT, which contains all the necessary data for the COSPAS-SARSAT protocol, must be completed and returned to the distributor so that the unit can be properly configured. For more information, refer to the Installation/Operation Manual of the ELT or contact your local registration authority.

7.2 ACTIVATION AND OPERATING MODES

FAMILIARIZATION WITH THE OPERATION

It is recommended to observe the following instructions to ensure reliable operation in the event of an emergency:

- (a) Become thoroughly familiar with the operation of the unit.
- (b) Always carry the Installation/Operation Manual of the ELT along with this supplement on board the aircraft during flight.
- (c) Visually inspect the unit at regular intervals, as specified in the Installation/Operation Manual. Check the ELT attachment, the antenna mounting and all cable connections for secure attachment.

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OPERATING MODES OF KANNAD 406 AF-COMPACT / INTEGRA 406 AF

The ELT is installed on the right side of the baggage compartment floor behind the copilot's seat. A remote control panel for the ELT is installed in the right section of the instrument panel above the engine instruments.

The following table provides an overview of the different operating modes of the ELT:

KANNAD 406 AF-COMPACT / INTEGRA 406 AF ELT				
Mode	Switch on ELT Unit/Remote Control Panel (RCP)	Function		
ARMED/ STANDBY	"ARM" (normal flight setting)	Stand-by mode for automatic activation of the ELT by the crash sensor (g-sensor). This mode is mandatory during flight. The switch on the ELT unit must be in the "ARM" position to allow operation of the ELT via the remote control panel.		
ON	"ON"	Overrides the crash sensor and activates ELT transmission manually (refer to the Installation Manual of the ELT for testing).		
OFF	"OFF" (ELT unit only)	Turns the ELT off for maintenance or to terminate the emergency signal transmission after rescue or inadvertent operation.		
RESET TEST	"RESET TEST" (remote control panel only)	To initiate the self-test function of the ELT and to terminate transmission of an activated ELT on the remote control panel.		

In order to be automatically activated by the crash sensor, the ELT must be in standby (**ARM**) mode. This mode is mandatory during flight. The ELT can only be operated with the remote control panel if it is in the stand-by mode (**ARM**). It is recommended to only switch the ELT OFF during maintenance or when the aircraft is parked for a longer period of time. Ensure that the ELT antenna is clear of obstructions.

In case of an emergency landing, it is recommended to tune in 121.5 MHz on the COM transceiver to check if the ELT has been activated. Once the ELT is activated, it can be manually deactivated by setting the ELT switch to the "OFF" position or by pressing the switch to the "RESET TEST" position on the ELT remote control panel for at least 1 second, and then returning the switch to the "ARMED" position. In the case of unintentional activation, national regulations with regard to informing Air Traffic Control must be observed.

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7.3 FUNCTIONAL TESTING

<u>GENERAL</u>

The ELT is furnished with a self-test function to perform an operational check to detect any possible malfunctions. An operational check using the self-test function must be conducted regularly by the pilot or maintenance personnel. The manufacturer recommends conducting a self-test once a month and after every system maintenance, but not more than once a week since every self-test consumes energy from the batteries. If the self-test is carried out more often than specified above, the battery lifetime is reduced accordingly. Functional and operational tests beyond the scope of a self-test, such as transmission tests, must be conducted by certified maintenance personnel in accordance with the procedures defined in the Installation Manual/Operation Manual/Inspection Log of the ELT. These types of tests must be conducted after the initial installation of the ELT as well as at regular intervals, according to national regulations.

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SELF-TEST PROCEDURE

CAUTION

Do not perform a self-test without the antenna connected because the transmitter could be damaged!

SELF-TEST KANNAD 406 AF-Compact / INTEGRA 406 AF ELT				
1	Set ELT switch to the "OFF" position	The ELT is installed on the right side of the baggage compartment floor behind the copilot's seat. The ELT switches to the OFF mode.		
2	Set ELT switch to the "ARM" position	A buzzer sounds during the whole self-test procedure. After a few seconds, the test result is displayed with the LED as follows:		
		• One long flash indicates that the system is operational and that no errors were found.		
		• A series of short flashes indicates that the test has failed and error conditions were found.		
3	Return the ELT switch to the "OFF" position or retain the "ARM" position	Setting the ELT switch back to the OFF position turns the ELT off. Before the next flight, the ELT must be switched to the ARMED mode (Stand-by mode).		

CAUTION

Provided that the ELT switch is in the ARM position, the self-test may also be initiated through the remote control panel by pushing the switch to the RESET TEST position. The self-test sequence is the same as described above for the ELT unit.

If the self-test fails, contact the manufacturer/distributor as soon as possible.

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Remark:

The number of flashes gives an indication to the fault detected during the self-test.

Number of flashes	FAILURE MODE
3+1	LOW BATTERY VOLTAGE
3+2	LOW RF POWER
3+3	FAULTY VCO LOCKING
3+4	NO IDENTIFICATION PROGRAMMED
3+5*	FAULTY ANTENNA CONNECTION
3+6*	FAULTY INTERNAL GPS

* INTEGRA 406 AF only

8. HANDLING, SERVICE AND MAINTENANCE

The ELT batteries must be replaced regularly according to the printed on expiry date, but also if e.g. an ELT activation has occured before. Refer to the Installation Manual/Operation Manual/Inspection Log of the ELT and the Maintenance Manual of the AQUILA AT01-100/200 for more details and a detailed maintenance schedule.

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